## Proposal

# Triple Modular Redundancy Implemented Through RISC-V Architecture on an FPGA

### PO# PC11801312

## by BYU-Idaho ECEN 499 W20 Senior Design Team 525 S Center ST, STC 320Y, Rexburg, Idaho 83460-1415

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#### Team members:

Max Bakes, Samuel Bagley, Michael Ashford, James Thomas, Zac Carico; Instructor: Dr. Kevin Smith

#### Description:

Create and test a system using Triple Modular Redundancy (TMR) with various configurations of the RISC-V open source instruction set on Microsemi's PolarFire radiation tolerant FPGA. This is to test the possibility of using RISC-V architecture in space, specifically the possibility of implementing this system in the Caution and Warning System (CWS) in the xEMU.

#### Methodology:

Altium will be used to design the PCB to breakout peripherals and include components that are listed in the hardware section of the document. The finished PCB design will be sent out to a company to print, and we will manually populate and solder the components.

To implement the TMR RISC-V cores, and program the PolarFire board, Microsemi's IDE will be used to create the VHDL code. Due to only having one license for this, Xilinx's Vivado will be used, along with Digilent's Basys3 board, will be used to design and test the communication hardware where possible and then transferred to the PolarFire once completed.

#### Hardware:

- PolarFire FPGA by Microsemi
- LCD Screen
- Radiation Sensor (Geiger Counter)
- Heartrate Monitor
- Pressure Sensor
- Temperature/Humidity Sensor

- Custom PCB with headers for GPIO, SPI, I2C, and UART. Ordered from China
- LEDs, switches, buzzers

Our custom PCB will house the LCD, the various sensors and components in use, and the connector that will attach onto the FPGA. We wanted to focus on monitoring astronaut vitals while in their suits. The heartrate monitor will look out for any elevations that could indicate problems. The temperature and humidity sensor will be used for temperature control in the suit. The pressure sensor will look out for suit leaks. The radiation sensor will monitor outside conditions and alert the astronaut to critical amounts of radiation. LEDs and buzzers will supplement the LCD screen in alerting the astronaut to danger.

#### **Deliverables:**

First deliverable will be this proposal. Afterwards, there will be a weekly status update on the project. At the end of the project, there will be the source code and documentation for the project, report on all research and testing done, JSC Form 290, completed PCB, and the PolarFire FPGA flashed with the final system design.

#### Schedule:

		WK1	WK2	WK3	WK4	WK5	WK6	WK7	WK8	WK9	WK10	WK11	WK12	WK13
		12-Jan-20	19-Jan-20	26-Jan-20	2-Feb-20	9-Feb-20	16-Feb-20	23-Feb-20	1-Mar-20	8-Mar-20	15-Mar-20	22-Mar-20	29-Mar-20	6-Apr-20
		c	c	C	c	c	c	c	c	c	C	C	C	5
Task	People Assigned	ß	ß	ß	Su	S	Su	ß	Su	ß	ß	ß	Su	f
PCB														
Design Multi-Channel ADC Circuit	Max				1.1	1.1	1.1							
LEDs (some RGB), Switches, GPIO Headers	James													
SPI & I2C Ports	James													
Radiation Sensor, Temperature and Humidity Sensor	Zac													
Heartrate Sensor, Pressure Sensor	Sam													
LCD Screen Interface (like the NHD- C0216CZ-FSW-FBW-3v3)	Michael													
UART Connector (50 pin)	James				1.1									
Design PCB Layout	Max, Sam, James													
Assemble PCB	Max, Sam, James													
COMM				1										
10 Full-DuplexUART (Using Hamming Code and FIFOs)	James													
SPI	Max						1.1							
12C	Sam													
PWM	Zac													
TMR/CPU														
Implement RISC-V cores with varying configurations	Michael, Zac													
RISC-V TMR	Michael, Zac					1.1								
FRAM TMR	Michael, Zac							1.0	1.1					
TESTING														
Write code to test system & peripherals	ALL											4		
Run Standard CPU Benchmarks	ALL													
Determine Ideal RISC-V Configuration	ALL													
DELIVERIES														
Poster														
Demonstration & Oral Report														1.1
Final Report														